

**EXAMPLE NO. 1**

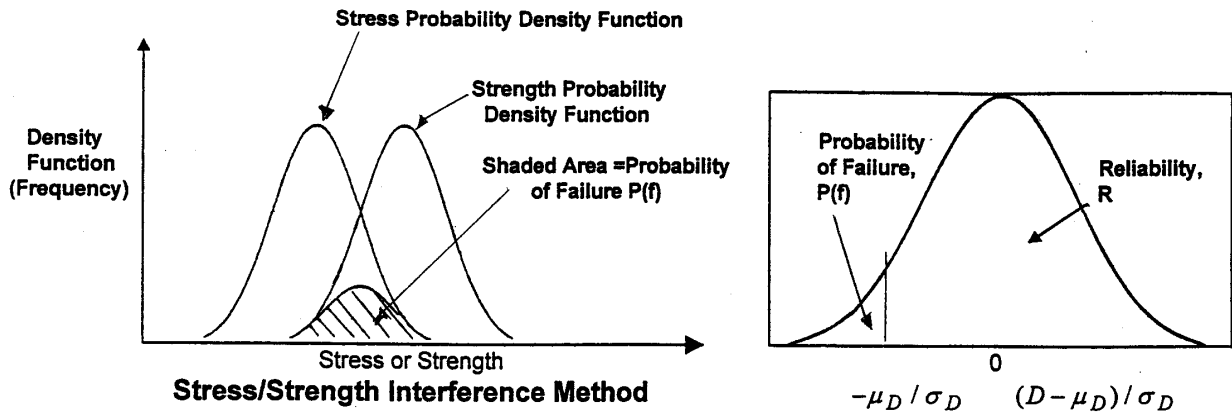
**MECHANICAL RELIABILITY**

*RADC-TR-85-194*

*October 1985*

**Stress/Strength Interference Method**

**for the  
Monopropellant Tank**



### Normalized Density Function of Excess Strength Over Load

Establish a new random variable  $D$ , where the difference of strength minus stress (load) is

$$D = \text{Strength (S)} - \text{Stress (L)} = S - L, \quad \mu_D = \mu_S - \mu_L, \quad \sigma_D = \sqrt{\sigma_S^2 + \sigma_L^2} \quad \text{and where}$$

$\mu_D, \mu_S, \mu_L$  = Mean of the difference, strength and stress, respectively, and

$\sigma_D, \sigma_S, \sigma_L$  = Standard Deviation of the difference, strength and stress, respectively.

Then the probability of failure,  $P(f)$ , is

$$P(f) = P(S - L < 0) = P(D < 0) = P\left\{\frac{D - \mu_D}{\sigma_D} < \frac{-\mu_D}{\sigma_D}\right\} = \frac{1}{\sqrt{2\pi}} \int_{-\infty}^{-\mu_D/\sigma_D} \exp(-t^2/2) dt$$

for the  $P(f) = \frac{1}{\sigma\sqrt{2\pi}} \int_{-\infty}^{+\infty} e^{-\frac{1}{2}\left(\frac{x-\mu}{\sigma}\right)^2} dx$  where the standard normal random variable at

$$t = \frac{x - \mu}{\sigma} \quad \& \quad -\mu_D / \sigma_D = -9.0766753347 \quad \text{at} \quad t = \frac{\mu_S - \mu_L}{\sqrt{\sigma_S^2 + \sigma_L^2}}, \quad \text{the difference probability density}$$

function is normalized to a zero mean and a standard deviation of one.

The reliability prediction,  $1 - P(f)$ , for the MTA Propellant Tank, which has a ultimate strength normal distribution mean of  $\mu_S = 160,000$  psi and a standard deviation of  $\sigma_S = 17,660$  psi and an in - orbit stress load (MEOP) normal distribution mean of  $\mu_L = 250$  psi and a standard deviation of  $\sigma_L = 50$  psi, is 0.999999999.

### Monopropellant Tank Reliability Mathematical Model and Prediction

## **EXAMPLE NO. 2**

### **FAILURE RATE ANALYSIS**

*Carderock Div, NSWC-92/L01,  
“Handbook of Reliability Prediction Procedures  
for Mechanical Equipment”, May 1992*

**for the  
Poppet Valve Assembly**

## Poppet Valve Assembly

$$\lambda_{PO} = \lambda_{PO,B} \frac{Q_a}{Q_f}$$

Where:

$\lambda_{PO}$  = Failure rate of the poppet assembly, failures/million operations

$\lambda_{PO,B}$  = Base failure rate for poppet assembly, failures/million operations

$Q_a$  = Leakage rate, in<sup>3</sup>/min

$Q_f$  = Leakage rate considered to be valve failure, in<sup>3</sup>/min

$$Q_a = \frac{2 \times 10^4 D_{MS} f^3 (P_1^2 - P_2^2)}{V_a L_W (S_S)^{3/2}}$$

Where:

$Q_a$  = Actual fluid leakage, in<sup>3</sup>/min

$D_{MS}$  = Mean seat diameter, in

$f$  = Mean surface finish of opposing surfaces, min

$P_1$  = Upstream pressure, lb/in<sup>2</sup>

$P_2$  = Downstream pressure, lb/in<sup>2</sup>

$V_a =$  Absolute fluid viscosity, lb-min/in<sup>2</sup>

$L_W =$  Radial seat land width, in.

$S_S =$  Apparent seat stress, lb/in<sup>2</sup>

$$\lambda_{PO} = \lambda_{PO,B} \cdot C_P \cdot C_Q \cdot C_F \cdot C_V \cdot C_N \cdot C_S \cdot C_{DT} \cdot C_{SW} \cdot C_W$$

Where:

$\lambda_{PO} =$  Failure rate of poppet assembly in failures/million operations; 1.26

$\lambda_{PO,B} =$  Base failure rate of poppet assembly, 1.40 failures/million operations

$C_P =$  Multiplying factor which considers the effect of fluid pressure on the base failure rate, 1.0

$C_Q =$  Multiplying factor which considers the effect of allowable leakage on the base failure rate, 1.0

$C_F =$  Multiplying factor which considers the effect of surface finish on the base failure rate, 1.0

$C_V =$  Multiplying factor which considers the effect of fluid viscosity/temperature on the base failure rate, 1.0

$C_N =$  Multiplying factor which considers the effect of contaminants on the base failure rate, 1.0625

$C_S$  = Multiplying factor which considers the effect of the apparent seat stress on the base failure rate, 0.621119

$C_{DT}$  = Multiplying factor which considers the effect of the seat diameter on the base failure rate, 1.09

$C_{SW}$  = Multiplying factor which considers the effect of the seat land width on the base failure rate, 1.001182

$C_W$  = Multiplying factor which considers the effect of flow rate on the base failure rate, 1.25

Where:

$$C_P = \left( \frac{P_1 - P_2}{3000} \right)^2$$

$$C_Q = 0.055 / Q_f \text{ For leakage (Per } GPM_R) > 0.03,$$

$$C_Q = 4.1 - (79Q_f) \text{ For leakage (Per } GPM_R) < 0.03,$$

$$C_F = \left( \frac{V_o}{V} \right)$$

Where:  $V_o = 2 \times 10^{-8} \text{ lb min / in}^2$

$$C_N \left( \frac{C_0}{C_{10}} \right)^3 N_{10} GPM_R$$

Where:

$$GPM_R = \text{Rated Flow in gallons/min, 5.0}$$

$$C_{10} = \text{Standard System Filter Size} = 10 \text{ micron}$$

$$C_0 = \text{System Filter Size in microns} = 5 \text{ micron}$$

$$N_{10} = 1.7 \text{ Particles under 10 microns/Hour/GPM}$$

$$C_S = \frac{1}{S_R^{3/2}} = 0.621119$$

Where:

$$S_R = \frac{12\pi D_M L_W}{D_S^2} = 0.758$$

$$S_S = \frac{P_S D_S^2}{4D_M L_W} = 1.2$$

$$S_S = \frac{\text{Force on Seat}}{\text{Seat Land Area}} = \frac{F_S}{A_{SL}}$$

$$F_S = \frac{\pi P_S D_S^2}{4}$$

$$\text{Stress Ratio} = S_C / S_S = S_R$$

Therefore, leakage varies with the seat stress as:

$$\left(\frac{1}{S}\right)^{3/2}$$

Minimum Contact Pressure =  $S_C = 3P_S$  approximately three times the fluid pressure.

$$A_{SL} = \pi D_M \cdot L_W$$

Where:

$A_{SL}$  = Seat land area, in<sup>2</sup>

$L_W$  = Land area width, in

$D_M$  = Mean land width diameter, in

$$A_{ST} = \frac{\pi(D_S)^2}{4}$$

Where:

$A_{ST}$  = Seat Area, in<sup>2</sup>

$D_S$  = Diameter of seat exposed to fluid pressure, Ps, 0.70 in

$$C_{DT} = 1.1 D_S + 0.32$$

$$C_{SW} = 3.55 - 24.52 L_W + 72.99 L_W^2 - 85.75 L_W^3 \text{ for } L_W < 6$$

$$C_W = 1 + \left[\frac{F_L}{100}\right]^2$$

Where:

$F_L$  = Ratio of actual flow rate to manufacturer's rating